AIRCRAFT TECHNICAL RECORD

Section 1. AIRFRAME LOG





DIAMOND AIRCRAFT INDUSTRIES INC. D.O.T. APPROVED MANUFACTURER #161-93 Registration:

N124SA

AC Serial Number:

40.NC242

Engine Serial Number: Prop Serial Number:

E4-A-06498

Civil Aviation Delegate

Airframe Total Time:

220559 3.2 HRS

Following limitations as per DA 40 AMM Chapter 4 (subject to change with revision):

Component	Hours	Years	Next Due:
Essential Bus	1000	-N/A-	1000 hrs
Pitot-Static System	1000	2	1000 hrs or MAY 2025
Bonding System	1000	-N/A-	1000 hrs
Emergency Battery Pack	-N/A-	2	APRIL 2025
ECU Backup Battery	-N/A-	1	MAY 2024

- The following STC's are incorporated:
 - FAA STC No. SA03366NY (Beringer wheel and brake system)
- THIS AIRCRAFT WAS INSPECTED FOR CONFORMITY TO APPLICABLE COMPANY RELEASED DOCUMENTS (DRAWINGS, SPECIFICATIONS, CONCESSIONS).
- I HEREBY CERTIFY THAT THIS AIRCRAFT CONFORMS TO FAA TYPE CERTIFICATE A47CE Rev No. 17, AND IS IN A FIT AND SAFE STATE FOR FLIGHT.
- Bi-weekly AWD check carried out to current revision 2023-10 all applicable AD's complied with, see attached.
- The work described above was carried out in accordance with the applicable Standard of Airworthiness.

Q 8	MAY 1 9 2023			
Inspector	Date			
(Intentionally left blank below this point)				

	,
U.S. Standard Airworthiness Certificate, issued:	
(Date) MAY 3 1 2023 , has been installed in the	
Aircraft on behalf of FAA Farmingdale/Saddle Brook MIDO Section AIR-8A5 on:	
MAY 2 1 2022	SPORT - IN
(Date)	Délégué de
(Print) R. Salton (Signature)	l'Avation civile

Section 1: Airframe Log

AIR TIME	LANDIN	IGS / CYCLES		
DATE DAILY TOTAL TOTAL	DAILY TOTAL	TOTAL	DETAILS OF TASK (INSPECTION, REPAIRS & REPLACE	



Aircraft Reg: N124SA Work Order: SO074233

AMO 161-93	Diamond Austro	DA40NG E4-A MTV-6-R/190-69	40.NC242 E4-A-06498 220559	Airframe TTSN	
Engine	MT	IVI I V-O 10 10 10		TSN:	
Propeller The following maintena				33	
The following maintena				35	

Aircraft stripe package installed. Aircraft stripe package installed.

Aircraft stripe package installed.

Rudder removed, rudder weight and residual moment verified within limits per DA40NG AMM Ch. 51-60-00 figure 2. Rudder reinstalled. and residual moment verified within moment and the flight controls affected by the work performed an independent inspection has been carried out on the flight controls affected by the work performed 80 THE MAINTENANCE DESCRIBED HAS BEEN PERFORMED IAW THE APPLICABLE STANDARDS OF AREA ORTHINESS PAGE TOTALS