

Aircraft Engine Test Verification

This document verifies that the engine model listed below has satisfactorily completed all testing listed below in accordance with TCM standard engine testing specifications as approved by the FAA.

Engine Model

IO550 N42B

Engine Serial Number

689998

Testing Completed

Standard Acceptance Test:

Date of Completion 11-15-06 Test Operator



Teledyne Continental Motors, Inc.
A Teledyne Technologies Company

Form No 98344

CONTINENTAL MOTORS

TCM NO: CYL0401



Teledyne Continental Motors, Inc.
A Teledyne Technologies Company

Engine Component Information Sheet

Printed: 11/16/2006

Serial: 689998
Spec: I0550N42B
New/Rebuilt: (NEW)

Assembled: 11/14/2006
Shipped:
Packed: 11/16/2006

Customer Name: CIRRUS DESIGN CORPORATION

Shipping Address:

Component	Serial Number
CAMSHAFT	Z06JA202
CRANKSHAFT	N06HA367
CRANKCASE	R06HA066
CONNROD	AE06IB633
CONNROD	AE06IB722
CONNROD	AE06IB660
CONNROD	AE06IB676
CONNROD	AE06IB635
CONNROD	AE06IB629
L MAGNETO	D06KA064
R MAGNETO	D06KA065
FUEL PUMP	B06JA262
MANIFOLD VALVE	C06JA205
METERING UNIT	A06JA254
STARTER	06 247 0024
ALTERNATOR	006JA147

Component	Serial Number
OIL COOLER	I06-4772-711
CYLINDER-1	AC05KA211
CYLINDER - 2	AC05KA800
CYLINDER - 3	AC05KA600
CYLINDER - 4	AC05KA231
CYLINDER - 5	AC05KA544
CYLINDER - 6	AC05KA579
NOZZLE - 1	1234
NOZZLE - 2	2234
NOZZLE - 3	3239
NOZZLE - 4	4236
NOZZLE - 5	5236
NOZZLE - 6	6229
PSTN WGT	WM



Pack Inspection Stamp

All of the information provided herein is subject to verification by the user. Teledyne Continental Motors, Inc. makes no representation or warranty concerning the accuracy or completeness of the information and assumes no responsibility with respect thereto.

TCM Engine Model	IO-550N42B
TCM Serial#	689998
Test Document	TH-3215 Rev G
Software Release#	1.98, 05/13/04
Start Time	11/15/06, 09:45:53
Accepted Time	11/15/06, 12:01:09
Cell# & Operator	TES, 8, 29274
Sea Level Power	310 HP @ 2700 Prop RPM
Vapor Pressure	0.75 In HG
Temp, Wet Baro	71.26 F, 29.52 In HG

Teledyne Continental Motors

Aircraft Engine Standard Acceptance Test Log

Test Club#	C5108A
Rqrd Pitch	20° ±.5
Actual Pitch	20°
Mixture Check	Pass
Alternator Check	PASS

Mag Data	Required	Actual
Eng Speed Both	2100	2112
Eng Speed Right	N/A	2047
Right Mag Drop	150	65
Eng Speed Left	N/A	2037
Left Mag Drop	150	75
Mag Drop Spread	50	10
Fuel Flow (Ref.)	62	68

Note: Magneto check between Run 2 & 3

Run Information	Run 1	Run 2	Run 3	Run 4	Run 5	Run 6	Run 7	Run 8
Time Of Day	09:53:47	09:55:11	09:59:20	10:05:05	10:15:13	10:31:53	10:40:09	11:47:28
Run Time Rqd (MM:SS)	01:00	01:00	01:00	10:00	10:00	01:00	10:00	05:00
Run Time	01:38	01:00	01:54	10:00	10:00	01:00	10:00	05:00
Prop Speed Rqd (RPM)	1175-1225	1575-1625	2425-2475	2719-2769	2485-2510	575-625	2479-2504	2477-2502
Prop Speed	1223	1594	2450	2746	2498	620	2489	2489
Manifold Press Rqd (InHg)	N/A	N/A	N/A	25.8-28.8	N/A	18.5 MAX	N/A	N/A
Manifold Press	11.5	13.2	20.9	27.5	23.0	12.5	22.5	22.4
Turbo Dis Press Rqd (InHg)	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Turbo Dis Press	28.9	28.9	28.8	28.8	28.8	28.8	28.8	28.8
Fuel Flow Rqd (lb/hr)	N/A	N/A	N/A	151.7-161.7	106.5-122.5	N/A	N/A	N/A
Fuel Flow	21.5	33.6	102.1	154.6	113.8	8.6	110.1	110.4
Nozzle Press Rqd (PSID)	N/A	N/A	N/A	19.9-20.5	12.4-13.6	N/A	N/A	N/A
Nozzle Press	4.5	5.0	12.1	20.4	13.3	4.0	12.9	12.9
Fuel Pump Press Rqd (PSIG)	N/A	N/A	N/A	30.0 REF	N/A	8.0-10.0	N/A	N/A
Fuel Pump Press	14.8	18.3	29.3	30.9	29.2	10.0	28.5	28.8
Fuel Temp Rqd (°F)	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Fuel Temp	70	71	75	79	83	79	76	73
Ambient Temp Rqd (°F)	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A
Ambient Temp	70	72	75	77	77	74	73	70
Eng Inlet Oil Temp Rqd (°F)	90-240	120-240	165-240	180-240	165-240	140-240	165-240	165-240
Eng Inlet Oil Temp	136	148	167	180	196	172	165	165
Eng Oil Press Rqd (PSIG)	30.0 MIN	30.0 MIN	30.0 MIN	49.6-60.6	30.0 MIN	18.7 MIN	30.0 MIN	30.0 MIN
Eng Oil Press	48.7	50.9	56.4	54.5	50.7	32.2	57.9	58.0
Min Cylinder Temp Rqd (°F)	150 MIN	175 MIN	250 MIN	250 MIN	250 MIN	N/A	250 MIN	250 MIN
Max Cylinder Temp Rqd (°F)	460 MAX	460 MAX	460 MAX	460 MAX	460 MAX	460 MAX	460 MAX	460 MAX
Cylinder 1 Temp	255	266	326	366	376	243	326	335
Cylinder 2 Temp	246	278	346	388	403	250	338	349
Cylinder 3 Temp	230	229	290	326	334	230	293	296
Cylinder 4 Temp	234	262	340	354	372	239	317	324
Cylinder 5 Temp	200	218	292	340	338	213	296	303
Cylinder 6 Temp	165	182	250	284	290	179	251	256
Collar Press Differential Rqd (PSID)	N/A	N/A	N/A	15.6 MAX	N/A	N/A	N/A	N/A
Collar Press Differential	1.3	1.1	0.8	0.2	0.1	-0.2	0.8	0.9



N335A

Continental Engine

Model: I.O.-550-N

Engine S/N. 689998

Engine Total Time 1302:5

Engine Time Since Major Overhaul

0:00

Engine Overhauled

Sept. 17, 2012

Under W.O. 212172

Done By

Zephyr Aircraft Engines Inc.

CRS # VI4R597M

39320 B Avenue

Zephyrhills, Florida 33542

Phone 813-788-3305

Fax 813-788-4412

ZEPHYR AIRCRAFT ENGINES INC.

CERTIFICATE of LIMITED WARRANTY

ZEPHYR AIRCRAFT ENGINES INC. (hereinafter Zephyr) makes the following warranty to it's customer subject to the limitations, conditions and exclusions set forth below.

WARRANTY COVERAGE-For a period of two years, or 500 hours following installation; Zephyr shall repair any aircraft engine which is found to be defective to mutual satisfaction, within said warranty period. After two years of operation, or 500 hours, and prior to manufacturers published TBO the cost of repairs including parts and labor shall be prorated in the following manner. The customer shall pay the net price for parts and labor used in effecting the repair divided by the applicable TBO, then multiplied by the number of hours on the engine which shall be deemed the greater of the actual hours in service of time at 40 hours per calendar month from the date of installation. The repair of any engine under the foregoing warranty shall not extend the period of warranty coverage set forth above.

CONDITIONS and PROCEDURES for OBTAINING WARRANTY WORK-The engine or component thereof may be returned prepaid to Zephyr's facility at Zephyrhills Florida with details in writing of the defect together with the log substantiation installation dates and time in service. Authorization must be obtained from Zephyr before repairs covered by warranty may be performed outside Zephyr's shop. In no event shall allowable labor costs exceed local published shops regular rates and Zephyr's time allowance. The foregoing warranty shall apply only if the engine has been properly installed and maintained in accordance with FAR's and manufacturers recommendations as specified in applicable factory service publications. The performance of said maintenance and inspections must be documented by logbook entries. Zephyr must be given notice within 10 day of discovery.

LIMITATIONS, EXCLUSIONS and DISCLAIMERS-Zephyr shall not assume freight charges, transportation or delivery expenses, installation or removal costs or airframe repairs all of which are excluded under the warranty. Zephyr does not warrant vendor items such as starters, magnetos, alternators, new cylinders, carburetors, fuel injectors, turbo system components, or crankcase and crankshaft repair as obtained from manufacturer or vendor. Zephyr will use it's best efforts to obtain warranty from the vendor of such items and pass all amounts recovered to customer. This warranty shall not apply to any engine which has been subject to misuse, neglect, accident, or operated in any manner which in the judgement of Zephyr has had an adverse affect on the engine or part. In the event of a dispute both parties shall agree to mediation prior to commencement of legal action.

THE FOREGOING WARRANTY IN LIEU OF ALL OTHER WARRANTIES OR REPRESENTATIONS, EXPRESS OR IMPLIED, ARISING BY OPERATION OF LAW OR OTHERWISE, INCLUDING WITHOUT LIMITATION THE WARRANTY OR MERCHANTABILITY OR FITNESS FOR PARTICULAR PURPOSE WHICH WARRANTIES ARE HEREBY EXCLUDED. Zephyr's liability hereunder shall be limited to the repair of any engine or part found to be defective within the applicable warranty period as set forth above. IN NO WAY SHALL ZEPHYR BE LIABLE FOR CONSEQUENTIAL OR INCIDENTAL DAMAGES OR ECONOMIC LOSS OF ANY NATURE WHETHER ARISING IN CONTRACT OR TORT, INCLUDING STRICT LIABILITY IN TORT OR NEGLIGENCE ON THE PART OF ZEPHYR.

Engine Mod.	Serial number	W.O. #	Date	Auth. Signature
I.O.-550-N	689998	212172	9/17/2012	<i>Heaven Volchok</i>

WO # 212172

PART NUMBER

CONTINENTAL IO-550-N

DESCRIPTION

ADVANCED AIRCRAFT SERVICES

QTY

64990	CRANKSHAFT	1	SN-N08GA109
IO-550	CRANKCASE REPAIR	1	BY DIVCO CN655435
SA646550-A1	GASKET SET	1	
641250	NOSE SEAL	1	
646592A2	MAIN BEARING SET	1	
642398	ROD BEARING	12	
530658	ROD BUSHING	6	
655958	ROD BOLTS	12	
654490	NUT ROD	12	
656990	BOLT CRK END GEAR	6	
654587-.44	BOLTS CAM	4	
649205	BOLTS ALT FACE	4	
641909	LOCK PLATES ALT FACE	2	
NAS1805-4	12 POINT NUTS	2	
641931-10.75	THRU BOLTS	8	
654084-J R	CAMSHAFT REGROUND	1	
SA646277	LIFTERS	6	
SA628488	LIFTERS	6	
X13041	BALL BEARING	1	
641368	NEEDLE BEARING	1	
501867	LOCK TAB	1	
MS35756-8	KEY	1	
630680	BEARING ROLLER	1	
534685	BALL BEARING	1	
521649-3	SNAP RING	1	
642335A1	SCREW KIT & PLUNGER	1	
634150	SPRING O-PRESS	1	
630046	PISTON PINS	6	
656642	ROCKER SHAFT	12	
501868	LOCK TABS	24	
655269	MAG BUSHING	4	
CH48109-1	OIL FILTER	1	
CAM539800M015	SPRING STARTER	1	
653470	GASKET STARTER	1	
652963	BUSHING ROCKER ARM	12	
653353-8A1	THROTTLE BODY O/H	1	S/N A06JA254
646433-9A4	MANIFOLD VALVE O/H	1	S/N C06JA205
655921-3A1	FUEL PUMP O/H	1	S/N B06JA262
654585 10281A	OIL COOLER O/H	1	S/N D11-9345-118
10-500556-1	MAG O/H	1	S/N D06KA064
10-500556-1	MAG O/H	1	S/N D06KA065
X1473	GASKET	6	
654439-2	HOSES INTAKE	6	
URHB32E	SPARK PLUG	12	
646644S4S7.00	HOSE	1	
646644S4Y10.62	HOSE	1	

646644R4S14.00	HOSE	1
655303	ID TAG TUNED FUEL INJ	6
10-423674-19	IGNITION HARNESS KIT	1

CYLINDER PARTS

636282O/S	GUIDE	6
655166O/S	GUIDE	6
AEC637781	VALVE	6
24802	KEY	24
654441	KEY	12
658842	SPRING	12
646985	SEAL	6
652112	ROTO COIL	6
654716P10	RING SET	6
654587P10	PISTON	6

AD NOTES COMPLIANCE RECORD

Engine - Teledyne Continental - IO-550-N


689998

Engine overhauled by Zephyr Aircraft Engines Inc. under W.O. #212172. Installed [6] O/H P010 cylinder asm. Installed new TCM Crankshaft & Damper Asm. P/N.649900 SN.N08GA109, Crankcase repaired by Divco, Camshaft reground, Fuel system O/H, Magneto's O/H, Oil cooler O/H, New Bendix harness and Tempest spark plugs. No other accessories done nor supplied by this agency at this time.

AD#	Applicable S.B. # & Subject	Date & Hours @ Comp.	Method of Compliance	One Time	Recurring	Next Comp. @ Hrs/Date	Authorized Signature & Number
88-03-06	Oil filter	09/12/2012 at O/H	Installed new oil filter P/N. CH48109-1	X			James H. Earnest Jr. VI4R597M
91-19-03	Champion oil filter	09/12/2012 at O/H	Installed new oil filter P/N. CH48109-1	X			James H. Earnest Jr. VI4R597M
93-08-17	Incorrect oil pick-up tube	09/12/2012 at O/H	N/A by Model per TCM S.B M91-10 rev.1	X			James H. Earnest Jr. VI4R597M
93-10-02	Cylinder valve retainer	09/12/2012 at O/H	Installed [6] O/H P010 cylinder assy. ALL keys in place.	X			James H. Earnest Jr. VI4R597M
93-16-16	Fuel pumps	09/12/2012 at O/H	N/A. By engine model & fuel pump P/N.	X			James H. Earnest Jr. VI4R597M
PL98-01-08	Exhaust roller rocker arms	09/12/2012 at O/H	N/A - not installed.	X			James H. Earnest Jr. VI4R597M
PL99-09-17	Crankshaft failure		Superseded by 99-19-01	X			James H. Earnest Jr. VI4R597M
99-19-01	Crankshaft failure	09/16/2008 at O/H	Installed New TCM crankshaft & damper assy. P/N.649900 S/N.N08GA109	X			James H. Earnest Jr. VI4R597M
PL2000-08-51	Crankshaft connecting rod journal fracture		Superseded by 2000-23-21	X			James H. Earnest Jr. VI4R597M
2000-23-21	Fracture of the crankshaft connecting rod journal	09/12/2012 at O/H	N/A. per TCM M.S.B. 00-5C by Date of crankshaft & engine S/N. . Crankshaft S/N.N08GA109 550	X			James H. Earnest Jr. VI4R597M
2004-08-10	Prevent loss of engine power	09/12/2012 at O/H	N/A. by cylinder manufacture.	X			James H. Earnest Jr. VI4R597M
2009-16-03	To prevent the separation of the cylinder head	09/12/2012 at O/H	N/A. by manufacture of cylinders installed on engine.	X			James H. Earnest Jr. VI4R597M
2009-24-52	To address rapid wear on the face of hydraulic lifters		Superseded by:2010-11-04	X			James H. Earnest Jr. VI4R597M
2010-11-04	To prevent excessive hydraulic lifter wear	09/12/2012 at O/H	N/A by P/N of lifters. Installed 6 New Superior P/N.SA646277 & 6 P/N.SA628488.	X			James H. Earnest Jr. VI4R597M
2012-03-06	To prevent an in-flight shutdown due to a failed fuel servo diaphragm	09/12/2012 at O/H	NA by type of fuel system.		X		James H. Earnest Jr. VI4R597M
2012-10-13	To prevent starter adapter gear shaft failure	09/12/2012 at O/H	NA by P/N. of starter adapter.	X			James H. Earnest Jr. VI4R597M

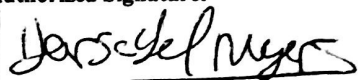
Zephyr Aircraft Engines Inc.
CRS #VI4R597M

James H. Earnest Jr.

1. Approving Civil Aviation Authority/Country Transport Canada		2. AUTHORIZED RELEASE CERTIFICATE FORM ONE			3. Form Tracking No. 12-098- 2/	
4. Organization Name and Address 978652 Ontario Ltd. Canadian Aero Manufacturing 2648 Ego Side Road, RR #3, Orillia, Ontario, Canada Tel: (705)326-1368 Fax: (705)326-8105 info@canadianaeromanufacturing.com Transport Canada Approval # 384 - 91					5. Work Order/Contract/Invoice 12-098	
6. Item 1	7. Description Starter Clutch Spring	8. Part No. CAM539800M15	9. Qty. 1	10. Serial/Batch No. 12-098- 2/	11. Status/work New	
12. Remarks AS PER DOT PDA # PDA01-11 DATED NOVEMBER 9,2001 ISSUE 2. This Part Complies With FAA TCDS # E253, E3CE, E1CE, E9CE, E273, 3E1, 3E3, E5CE, E8CE, E3S0 This TCCA-PDA Part is not a critical component						
13a. Certifies that the items identified above were manufactured in conformity to:			14a.			
<input checked="" type="checkbox"/> approved design data and are in condition for safe operation. <input type="checkbox"/> non approved design data specified in block 12.			<input type="checkbox"/> CARSTEDT Maintenance Release <input type="checkbox"/> Other: _____			
13b. Signature 		13c. Approved Organization Number 384-91		14b. Signature N/A		14c. Approved Organization Number N/A
13d. Name Joe Frantz		13e. Date (dd/mmm/yyyy) 13/Apr/2012		14d. Name N/A		14e. Date (dd/mmm/yyyy) N/A

Installer Responsibilities

This certificate does not constitute authority to install.
Installers working in accordance with the national regulations of a country other than the specified in block 1 must ensure that their regulations recognize certifications from the country specified.
Statements in blocks 13a or 14a do not constitute installation certification. In all cases, the technical record of the aircraft must contain an installation certification issued in accordance with the applicable national regulations before the aircraft may be flown.

1. Approving National Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number: WDC # 107811	
4. Organization Name and Address: DIVCO, INC. 2806 N. SHERIDAN RD TULSA, OK 74115						5. Work Order/Contract/Invoice Number: INVOICE #102444	
6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:	
1	CONTINENTAL IO-550 CRANKCASE	CASTING #655435	N/A	1 EACH	MATCH #R06HA066	REPAIRED	
13. Remarks: CUT MATING SURFACES AND RESIZED BORES TO MANUFACTURER'S LIMITS. CRANKCASE REPAIRED PER DIVCO APPROVED PROCESS SPECS 77-002, REV. AUGUST 21, 2007. DATE COMPLETED: 08/15/2012 WDC # 107811 CERTIFIES WORK SPECIFIED IN BLOCKS 12/13 WAS CARRIED OUT IN ACCORDANCE WITH EASA PART 145 AND WITH RESPECT TO THAT WORK THE COMPONENT IS CONSIDERED READY FOR RELEASE TO SERVICE UNDER EASA PART 145 APPROVAL NUMBER EASA 145-528							
14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.				19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
15. Authorized Signature:		16. Approval/Authorization No.:		20. Authorized Signature: 		21. Approval/Certificate No.: DB2R762K	
17. Name (Typed or Printed):		18. Date (m/d/y):		22. Name (Typed or Printed): HERSCHEL MYERS		23. Date (m/d/y): AUG 15 2012	
User/Installer Responsibilities							
It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1. Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.							

DIVCO CRANKCASE WORK ORDER

WDC NO. 107811 DATE RECEIVED 7/30/12
 Customer Zephyr A/C Engines P.O. No. 212172
 Address Zephyrhills, FL Carrier UPS
 Phone 813-988-3305 Model IO-550 C/N. 1055435
 Case Mfg. CONF. S.N. _____ Match No. RobtH A06b

Customer Orders and Discrepancies

RECEIVING/REPAIR SEQUENCE	DATE	MECH. I.D.
1. Preliminary Insp. clean and Prep. Includes removal of LYC anchored Thru bolts	7/30	AA
2. Record Previous Repair W.O.s		
3. Record Deck Height LH <u>560</u> RH <u>560</u>	8/6	AB
4. NDI (HIDDEN DAMAGE) Mark Discontinuities (go to step 8 if weld not required)		
5. Preheat and Arc Insp. (min. 400 F)		
6. Prep. For Welding and weld (Note Weld Areas in description of repair)	8/8	AB
7. Post Heat (4 Hrs. 400 F) Insp. For Crack Indications Visual		
8. Relocate Stud/Through Bolt Holes	8/7	CI
9. Mach. Cyl. Bore Surface		
10. Misc Deck Work		
11. Cut Mating Surface	8/7	JH
12. Rough Machine Bearing Bores (if required)		
13. If applicable install LYC anchored thru bolts Assemble/Torque	8/8	BO

Description of Repair

Cut mating surfaces and
 resized bores to factory
 limits.

SERVICEABLE

Reject Remarks

Maint. Rel./8130-3 Issued by J. S. Myers

Dated

8-15-12
PH

214682



ZEPHYR AIRCRAFT ENGINES, INC

FAA CERTIFIED REPAIR STATION # VI4R597M

39320 B AVE
813-788-3305

ZEPHYRHILLS, FL 33542
FAX 813-788-4412

W.O.# 212172

DATE: 09/13/2012

ADVANCED AIRCRAFT SERVICES
530 AIRPORT RD
LITITZ, PA 17543

DESCRIPTION OF SERVICES:

CONTINENTAL IO-550-N ENGINE SN # 689998 DISASSEMBLED, ALL PARTS CLEANED AND INSPECTED, STEEL PARTS MAGNA FLUXED ENGINE REASSEMBLED MAJOR OVERHAULED TO FACTORY NEW TOLERANCES AND TEST RUN.

MAINTENANCE RELEASE

THIS AIRCRAFT , AIRCRAFT ENGINE (OR APPLIANCE) IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT MAINTENANCE RULES OF THE FEDERAL AVIATION REGULATIONS AND IS APPROVED FOR RETURN TO SERVICE PERTINENT DETAILS OF THIS REPAIR ARE ON FILE AT THIS AGENCY UNDER W.O.# 212172

DATE: 09/13/2012

W.O.# 212172

FOR ZEPHYR AIRCRAFT ENGINES INC.
39320 B AVE, ZEPHYRHILLS, FL 33542
FAA CERTIFIED REPAIR STATION #VI4R597M

SIGNED: 

PRICE PER QUOTE
NEW CRANKSHAFT

\$26,900.00
\$7,130.00

SUBTOTAL: \$34,030.00
LESS DEPOSIT \$0.00
SUBTOTAL: \$34,030.00
SALES TAX: 6%
SALES TAX: 1% DISCRETIONARY TAX ON \$5,000.00
FREIGHT SHIPPED TO PA 1WAY FRT \$200.00
TOTAL BALANCE DUE: \$34,230.00